

Physics 601 Homework 5---Due Friday October 8

1. A standard result in undergraduate relativity is the velocity addition formula:

$$v = \frac{v_1 + v_2}{1 + v_1 v_2}.$$

This typically obtained by taking the product of two Lorentz transformations and the doing some algebra. A more straightforward way to

obtain this use the four velocity: $u^\mu = \begin{pmatrix} \gamma \\ \gamma v_x \\ \gamma v_y \\ \gamma v_z \end{pmatrix}$ $\gamma = \frac{1}{\sqrt{1 - \vec{v}^2}}$. Suppose I start with a

particle moving with velocity \vec{v}_1 with an associated four-velocity $u_1^\mu = \begin{pmatrix} \gamma_1 \\ \gamma_1 v_{1x} \\ \gamma_1 v_{1y} \\ \gamma_1 v_{1z} \end{pmatrix}$.

Suppose one boosts to a new frame by running to the left (-x direction) with a

velocity which corresponds to a Lorentz transformation $\Lambda^\mu_\nu = \begin{pmatrix} \gamma_2 & v_2 \gamma_2 & 0 & 0 \\ v_2 \gamma_2 & \gamma_2 & 0 & 0 \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \end{pmatrix}$.

The four-velocity in the new frame is $u^\nu = \Lambda^\mu_\nu u_1^\mu$.

- a. From the transformed four velocity find v_x, v_y, v_z in the new frame.

- b. For the case where \vec{v}_1 is entirely along the x direction, show that $v = \frac{v_1 + v_2}{1 + v_1 v_2}$.

2. Consider relativistic transformations restricted to one spatial direction. In that case, the velocity can be specified by a single number from -1 to 1 (in units with $c=1$). It is convenient to introduce the "rapidity" η with the property that $v = \tanh(\eta)$. Note that while v is restricted from -1 to 1, η goes from $-\infty$ to ∞ .

- a. Show that the 4-velocity is given by $u^\mu = \begin{pmatrix} \cosh(\eta) \\ \sinh(\eta) \\ 0 \\ 0 \end{pmatrix}$.

- b. Show that while velocities do not add linearly in relativity rapidities do. That is one obtains the relativistic velocity addition formula by taking $\eta = \eta_1 + \eta_2$. In a certain sense this turns out to be just the hyperbolic trig version of the angle addition formula for two-dimensional rotations.

3. Relativistic Kinematics: Consider the elastic scattering of two particles with masses m_1 and m_2 . Suppose initially particle 1 is at rest and particle 2 approaches it

$$1. u^\mu = \begin{pmatrix} \gamma \\ \gamma v_x \\ \gamma v_y \\ \gamma v_z \end{pmatrix}, \quad \gamma = \frac{1}{\sqrt{1-v^2}}$$

start with a particle moving w/ \vec{v}_1 , w/ $u_1^\mu = \begin{pmatrix} \gamma_1 \\ \gamma_1 v_{1x} \\ \gamma_1 v_{1y} \\ \gamma_1 v_{1z} \end{pmatrix}$

boost to new frame, $\Lambda_V^\mu = \begin{bmatrix} \gamma_2 & v_2 \gamma_2 & 0 & 0 \\ v_2 \gamma_2 & \gamma_2 & 0 & 0 \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix}$
 (by moving to left)

four velocity in new frame $= u^\nu = \Lambda_V^\mu u_1^\nu$

a) From u^ν , find v_x, v_y, v_z in new frame

$$u^\nu = \begin{bmatrix} \gamma_2 & v_2 \gamma_2 & 0 & 0 \\ v_2 \gamma_2 & \gamma_2 & 0 & 0 \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} \gamma_1 \\ \gamma_1 v_{1x} \\ \gamma_1 v_{1y} \\ \gamma_1 v_{1z} \end{bmatrix} = \begin{bmatrix} \gamma_1 \gamma_2 + \gamma_1 \gamma_2 v_{1x} v_2 \\ \gamma_1 \gamma_2 v_2 + \gamma_1 \gamma_2 v_{1x} \\ \gamma_1 v_{1y} \\ \gamma_1 v_{1z} \end{bmatrix}$$

new frame: $\gamma = \gamma_1 \gamma_2 + \gamma_1 \gamma_2 v_{1x} v_2$

$$\gamma v_x = \gamma_1 \gamma_2 v_2 + \gamma_1 \gamma_2 v_{1x}$$

$$v_x = \frac{\gamma_1 \gamma_2 v_2 + \gamma_1 \gamma_2 v_{1x}}{\gamma_1 \gamma_2 + \gamma_1 \gamma_2 v_{1x} v_2} = \frac{v_2 + v_{1x}}{1 + v_{1x} v_2} = v_x$$

$$v_y = \frac{v_{1y}}{\gamma_2 (1 + v_{1x} v_2)}$$

$$v_z = \frac{v_{1z}}{\gamma_2 (1 + v_{1x} v_2)}$$

b) show if $\vec{v}_1 = v_{1x} \hat{x}$, $v = \frac{v_1 + v_2}{1 + v_1 v_2}$

$$\therefore v_{1y} = v_{1z} = 0$$

$$u_1^\mu = \begin{pmatrix} \gamma_1 \\ \gamma_1 v_{1x} \\ 0 \\ 0 \end{pmatrix} \rightarrow u^\nu = \begin{bmatrix} \gamma_1 \gamma_2 (1 + v_1 v_2) \\ \gamma_1 \gamma_2 (v_2 + v_1) \\ 0 \\ 0 \end{bmatrix} \quad v = \sqrt{v_x^2 + v_y^2 + v_z^2} = v_x \quad \text{b/c } v_y = v_z = 0$$

$$v = \frac{v_2 + v_1}{1 + v_1 v_2} \quad \checkmark$$

2. η = "rapidity", $v = \tanh(\eta)$

$c=1$, so $-1 \leq v \leq 1 \rightarrow -\infty < \eta < \infty$

(restricted to one spacial direction)

a) Show $u^\mu = \begin{bmatrix} \cosh(\eta) \\ \sinh(\eta) \\ 0 \\ 0 \end{bmatrix}$, $u^\mu = \begin{pmatrix} \gamma \\ \gamma v_x \\ \gamma v_y \\ \gamma v_z \end{pmatrix}$

b/c $v_y = v_z = 0$, $|\vec{v}| = v_x$

$$\gamma = \frac{1}{\sqrt{1-v^2}} \quad v^2 = \tanh^2(\eta)$$

$$1 - \tanh^2(\eta) = \operatorname{sech}^2 \eta$$

$$\gamma = \frac{1}{\operatorname{sech}^2 \eta} = \cosh \eta$$

$$v_x = \tanh \eta, \quad \gamma v_x = \tanh \eta \cosh \eta = \sinh \eta$$

$$\therefore u^\mu = \begin{bmatrix} \cosh \eta \\ \sinh \eta \\ 0 \\ 0 \end{bmatrix} \quad \checkmark$$

b) Show that while velocities do not add linearly, rapidities do.

$$\left(v = \frac{v_1 + v_2}{1 + v_1 v_2} \right) \quad v_1 = \tanh(\eta_1) \quad v_2 = \tanh(\eta_2)$$

$$\eta = \eta_1 + \eta_2 = \tanh^{-1}(v)$$

$$v = \tanh(\eta) = \tanh(\eta_1 + \eta_2)$$

Schaum's: $\tanh(x+y) = \frac{\tanh x + \tanh y}{1 + \tanh x \tanh y}$

$$v = \frac{\tanh \eta_1 + \tanh \eta_2}{1 + \tanh \eta_1 \tanh \eta_2} = \frac{v_1 + v_2}{1 + v_1 v_2} \quad \checkmark$$

4. Rocket firing at constant rate
rest frame it experiences acceleration in +x direction

10

a) Show that EOM: $\frac{du^\mu}{d\tau} = G^{\mu\nu} u_\nu$, $G^{\mu\nu} = \begin{pmatrix} 0 & -g & 0 & 0 \\ g & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \end{pmatrix}$

1st show acceleration in comoving is correct

$$a^\mu = G^{\mu\nu} u_\nu = \begin{bmatrix} 0 & -g & 0 & 0 \\ g & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix} \begin{bmatrix} \gamma \\ -\gamma v_x \\ -\gamma v_y \\ -\gamma v_z \end{bmatrix} = \begin{bmatrix} g\gamma v_x \\ g\gamma \\ 0 \\ 0 \end{bmatrix} = g\gamma \begin{bmatrix} v_x \\ 1 \\ 0 \\ 0 \end{bmatrix}$$

comoving frame

moving in proper time, $\frac{du^\mu}{d\tau} = \frac{du^\mu}{dt}$, $v \ll 1 \rightarrow u^\mu = \begin{bmatrix} 1 \\ v_x \\ v_y \\ v_z \end{bmatrix} \rightarrow \frac{du^\mu}{dt} = \begin{bmatrix} 0 \\ a \\ 0 \\ 0 \end{bmatrix}$
b/c $v_{rocket} \rightarrow 0$ in comoving frame such that $\gamma \sim 1$

$$G^{\mu\nu} u_\nu = \begin{bmatrix} 0 & -g & 0 & 0 \\ g & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix} \begin{bmatrix} 1 \\ 0 \\ 0 \\ 0 \end{bmatrix} = \begin{bmatrix} 0 \\ g \\ 0 \\ 0 \end{bmatrix}$$

so, $\frac{du^\mu}{d\tau} = G^{\mu\nu} u_\nu \rightarrow a = g$ in comoving frame ✓

$$\frac{d(u_\mu u^\mu)}{d\tau} = \left(\frac{du_\mu}{d\tau}\right) u^\mu + u_\mu \left(\frac{du^\mu}{d\tau}\right) = a_\mu u^\mu + u_\mu a^\mu$$

$$a^\mu = \begin{bmatrix} g\gamma v_x \\ g\gamma \\ 0 \\ 0 \end{bmatrix}, \quad a_\mu = \begin{bmatrix} g\gamma v_x \\ -g\gamma \\ 0 \\ 0 \end{bmatrix}, \quad u^\mu = \begin{bmatrix} \gamma \\ \gamma v_x \\ \gamma v_y \\ \gamma v_z \end{bmatrix}, \quad u_\mu = \begin{bmatrix} \gamma \\ -\gamma v_x \\ -\gamma v_y \\ -\gamma v_z \end{bmatrix}$$

$$\frac{d(u_\mu u^\mu)}{d\tau} = (g\gamma^2 v_x - g\gamma^2 v_x) + (\gamma^2 g v_x - \gamma^2 g v_x) = 0 \quad \checkmark$$

b) Starts at origin at $t=0$

find expression for $u^\mu(\tau)$

$$\frac{du^\mu}{d\tau} = a^\mu, \text{ split into components } \rightarrow$$

$$\frac{du^0}{d\tau} = a^0 = g u^1$$

$$\frac{du^2}{d\tau} = a^2 = 0$$

$$\frac{du^1}{d\tau} = a^1 = g u^0$$

$$\frac{du^3}{d\tau} = a^3 = 0$$

4 cont

$$\frac{d}{d\tau} \left(\frac{du'}{d\tau} = g u^0 \right) \Rightarrow \frac{d^2 u'}{d\tau^2} = g \frac{du^0}{d\tau} = g^2 u'$$

$$\frac{d^2 u'}{d\tau^2} - g^2 u' = 0$$

has solutions of the form $u' = A e^{g\tau} + B e^{-g\tau}$

$$u' = A \sinh g\tau + B \cosh g\tau$$

because the rocket starts from rest at $t=0$ at origin
 $\tau=0$

$$u'(\tau=0) = A \sinh(0) + B \cosh(0) = B \quad \therefore B=0$$

$$u'(\tau) = A \sinh g\tau$$

$$g u^0 = \frac{du'}{d\tau} = A g \cosh g\tau \quad \rightarrow u^0(\tau) = A \cosh g\tau$$

u^2 and $u^3 = \text{constants}$, but b/c of the I.C. $u^2 = u^3 = 0$

$$u^\mu = \begin{bmatrix} A \cosh g\tau \\ A \sinh g\tau \\ 0 \\ 0 \end{bmatrix}, \quad u^\mu u_\mu = A^2 \cosh^2 g\tau - A^2 \sinh^2 g\tau = A^2 = 1$$

$$A = \pm 1$$

$A = +1$ b/c the acceleration is in the +x direction

$$u^\mu(\tau) = \begin{bmatrix} \cosh g\tau \\ \sinh g\tau \\ 0 \\ 0 \end{bmatrix}$$

c) compute $x^\mu(\tau)$

$$\frac{dx^0}{d\tau} = \cosh g\tau$$

$$\frac{dx^1}{d\tau} = \sinh g\tau$$

$$\frac{dx^2}{d\tau} = \frac{dx^3}{d\tau} = 0$$

$$\downarrow$$
$$x^0 = \frac{\sinh g\tau}{g} + c_0$$

$$\downarrow$$
$$x^1 = \frac{\cosh g\tau}{g} + c_1$$

$$x^2 = c_2 \quad x^3 = c_3$$

4 cont

apply I.C.

$$x^0(\tau=0) = 0 \rightarrow c_0 = 0$$

$$x^2(\tau=0) = 0 \rightarrow c_2 = 0$$

$$x^3(\tau=0) = 0 \rightarrow c_3 = 0$$

$$x'(\tau=0) = 0 = \frac{1}{g} + c_1 \rightarrow c_1 = -\frac{1}{g}$$

$$x^u(\tau) = \frac{1}{g} \begin{bmatrix} \sinh g\tau \\ \cosh g\tau - 1 \\ 0 \\ 0 \end{bmatrix}$$

d) Find $x(t)$

from part c, $t(\tau) = \frac{1}{g} \sinh g\tau$

$$gt = \sinh g\tau$$

$$\tau = \frac{\sinh^{-1}(gt)}{g}$$

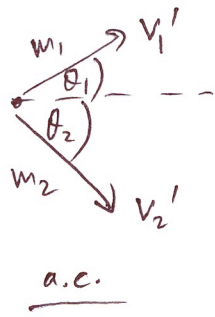
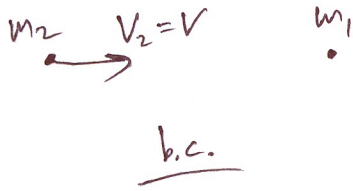
$$x(\tau) = \frac{1}{g} (\cosh g\tau - 1)$$

↓

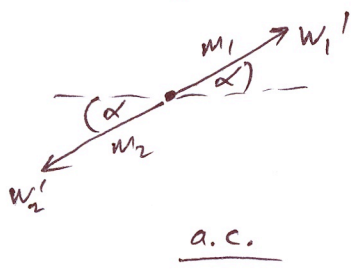
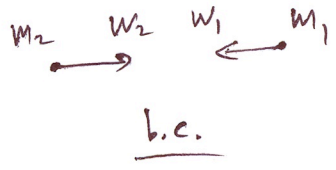
$$x(t) = \frac{1}{g} (\cosh(\sinh^{-1}(gt)) - 1)$$

$$= \frac{1}{g} [\sqrt{1+g^2t^2} - 1]$$

3



L → z



Define

$$v = \tanh \eta$$

$$v_{cm} = \tanh \eta_{cm}$$

$$v_1' = \tanh \eta_1'$$

$$v_2' = \tanh \eta_2'$$

$$v_1' \cos \theta_1 = \tanh \bar{\lambda}_1$$

$$v_1' \sin \theta_1 = \tanh \bar{\lambda}_1$$

$$v_2' \cos \theta_2 = \tanh \bar{\lambda}_2$$

$$v_2' \sin \theta_2 = \tanh \bar{\lambda}_2$$

$$w_1 = \tanh \xi_1$$

$$w_2 = \tanh \xi_2$$

$$w_1' = \tanh \xi_1'$$

$$w_2' = \tanh \xi_2'$$

$$w_1' \cos \alpha = \tanh \mu_1$$

$$w_1' \sin \alpha = \tanh \bar{\mu}_1$$

$$w_2' \cos \alpha = \tanh \mu_2$$

$$w_2' \sin \alpha = \tanh \bar{\mu}_2$$

Find v_{cm} . Set $c=1$.

Look the particles before collision. In lab frame, the total energy and total momentum are $E = m_1 + \gamma(v)m_2$ and $P = \gamma(v)m_2v$. In CM frame, the total spatial momentum is zero. So, using Lorentz transformation,

$$0 = -\gamma(v_{cm}) v_{cm} E + \gamma(v_{cm}) P$$

$$v_{cm} = \frac{P}{E}$$

$$v_{cm} = \frac{\gamma(v)m_2v}{m_1 + \gamma(v)m_2}$$

Addition of velocities

$$\left. \begin{aligned} \eta &= \xi_2 + \eta_{cm} \\ 0 &= -\xi_1 + \eta_{cm} \end{aligned} \right\} \text{b.c.}$$

$$\left. \begin{aligned} \lambda_1 &= \mu_1 + \eta_{cm} \\ \lambda_2 &= -\mu_2 + \eta_{cm} \end{aligned} \right\} \begin{array}{l} \text{a.c.} \\ z\text{-direction} \end{array}$$

$$\left. \begin{aligned} \bar{\lambda}_1 &= \bar{\mu}_1 \\ \bar{\lambda}_2 &= \bar{\mu}_2 \end{aligned} \right\} \begin{array}{l} \text{a.c.} \\ \text{transverse direction} \end{array}$$

Find w_1, w_2 .

$$w_1 = \tanh \xi_1 = \tanh \eta_{cm} = v_{cm}$$

$$\boxed{w_1 = \frac{\gamma(v) m_2 v}{m_1 + \gamma(v) m_2}}$$

$$w_2 = \tanh \xi_2 = \tanh (\eta - \eta_{cm}) = \frac{\tanh \eta - \tanh \eta_{cm}}{1 - \tanh \eta \cdot \tanh \eta_{cm}} = \frac{v - v_{cm}}{1 - v v_{cm}}$$

$$\boxed{w_2 = \frac{\gamma(v) m_1 v}{\gamma(v) m_1 + m_2}}$$

So basically we have written w_1, w_2 and v_{cm} as the functions of v .
Then it is sufficient if we express the next variables in terms of w_1, w_2 and v_{cm} .

Note that if $m_1 = m_2$, $w_1 = w_2$.

Find w_1', w_2' .

We work in CM frame, and use conservation of energy and momentum.

$$\gamma(w_1) m_1 + \gamma(w_2) m_2 = \gamma(w_1') m_1 + \gamma(w_2') m_2$$

$$\gamma(w_1) m_1 w_1 = \gamma(w_2) m_2 w_2$$

$$\gamma(w_1') m_1 w_1' = \gamma(w_2') m_2 w_2'$$

In rapidity notation,

$$m_1 \cosh \xi_1 + m_2 \cosh \xi_2 = m_1 \cosh \xi_1' + m_2 \cosh \xi_2'$$

$$m_1 \sinh \xi_1 = m_2 \sinh \xi_2$$

$$m_1 \sinh \xi_1' = m_2 \sinh \xi_2'$$

The last two equations, however, can be written as

$$m_1 w_1 \cosh \xi_1 = m_2 w_2 \cosh \xi_2$$

$$m_1 w_1' \cosh \xi_1' = m_2 w_2' \cosh \xi_2'$$

So, substituting to the first equation,

$$m_1 \cosh \xi_1 + m_1 \frac{w_1}{w_2} \cosh \xi_1 = m_1 \cosh \xi_1' + m_1 \frac{w_1'}{w_2'} \cosh \xi_1'$$

But,

$$\frac{w_1'}{w_2'} = \frac{m_2}{m_1} \frac{\cosh \xi_2'}{\cosh \xi_1'} = \frac{m_2}{m_1} \frac{\gamma(w_2')}{\gamma(w_1')} = \frac{m_2}{m_1} \sqrt{\frac{1-(w_1')^2}{1-(w_2')^2}}$$

$$\left(\frac{w_1'}{w_2'}\right)^2 = \left(\frac{m_2}{m_1}\right)^2 \left(\frac{1-(w_1')^2}{1-(w_2')^2}\right)$$

$$\left(\frac{w_1'}{w_2'}\right)^2 - (w_1')^2 = \left(\frac{m_2}{m_1}\right)^2 (1-(w_1')^2)$$

$$\left(\frac{w_1'}{w_2'}\right)^2 = (w_1')^2 + \left(\frac{m_2}{m_1}\right)^2 (1-(w_1')^2)$$

Substitute it back, we'll get

$$m_1 \cosh \xi_1 \left(1 + \frac{w_1}{w_2}\right) = m_1 \cosh \xi_1' \left(1 + \sqrt{(w_1')^2 + \left(\frac{m_2}{m_1}\right)^2 (1-(w_1')^2)}\right)$$

$$\gamma(w_1) \left(1 + \frac{w_1}{w_2}\right) = \gamma(w_1') \left(1 + \sqrt{\gamma(w_1')^2 \left(1 - \frac{1}{\gamma(w_1')^2}\right) + \left(\frac{m_2}{m_1}\right)^2}\right)$$

$$\gamma(w_1) \left(1 + \frac{w_1}{w_2}\right) = \gamma(w_1') + \sqrt{\gamma(w_1')^2 + \left[\left(\frac{m_2}{m_1}\right)^2 - 1\right]}$$

$$\gamma(w_1) \left(1 + \frac{w_1}{w_2}\right) - \gamma(w_1') = \sqrt{\gamma(w_1')^2 + \left[\left(\frac{m_2}{m_1}\right)^2 - 1\right]}$$

$$\gamma(w_1)^2 \left(1 + \frac{w_1}{w_2}\right)^2 + \gamma(w_1')^2 - 2\gamma(w_1)\gamma(w_1') \left(1 + \frac{w_1}{w_2}\right) = \gamma(w_1')^2 + \left[\left(\frac{m_2}{m_1}\right)^2 - 1\right]$$

$$\gamma(w_1)^2 \left(1 + \frac{w_1}{w_2}\right)^2 - \left[\left(\frac{m_2}{m_1}\right)^2 - 1\right] = 2\gamma(w_1)\gamma(w_1') \left(1 + \frac{w_1}{w_2}\right)$$

$$\boxed{\gamma(w_1') = \frac{1}{2} \gamma(w_1) \left(1 + \frac{w_1}{w_2}\right) - \frac{1}{2\gamma(w_1)} \left(1 + \frac{w_1}{w_2}\right)^{-1} \left[\left(\frac{m_2}{m_1}\right)^2 - 1\right]}$$

Note that if $m_1 = m_2$, $\gamma(w_1') = \gamma(w_1)$.

Back to our first equation (the conservation of energy),

$$\gamma(w_1) m_1 + \gamma(w_2) m_2 = \gamma(w_1') m_1 + \gamma(w_2') m_2$$

$$= \frac{1}{2} m_1 \gamma(w_1) \left(1 + \frac{w_1}{w_2}\right) - \frac{m_1}{2\gamma(w_1)} \left(1 + \frac{w_1}{w_2}\right)^{-1} \left[\left(\frac{m_2}{m_1}\right)^2 - 1\right] + \gamma(w_2') m_2$$

$$\frac{m_1}{m_2} \gamma(w_1) + \gamma(w_2) = \frac{1}{2} \frac{m_1}{m_2} \gamma(w_1) \left(1 + \frac{w_1}{w_2}\right) - \frac{1}{2\gamma(w_1)} \frac{m_1}{m_2} \left[\left(\frac{m_2}{m_1}\right)^2 - 1\right] \left(1 + \frac{w_1}{w_2}\right)^{-1} + \gamma(w_2')$$

$$\boxed{\gamma(w_2') = \frac{1}{2} \frac{m_1}{m_2} \gamma(w_1) \left(1 - \frac{w_1}{w_2}\right) + \gamma(w_2) + \frac{1}{2\gamma(w_1)} \frac{m_1}{m_2} \left[\left(\frac{m_2}{m_1}\right)^2 - 1\right] \left(1 + \frac{w_1}{w_2}\right)^{-1}}$$

If $m_1 = m_2$, $\gamma(w_2') = \gamma(w_2)$.

Find v_1' .

$$v_1' \sin \theta_1 = \tanh \bar{\lambda}_1 = \tanh \bar{\mu}_1 = w_1' \sin \alpha \implies \sin \alpha = \frac{v_1'}{w_1'} \sin \theta_1$$

$$v_1' \cos \theta_1 = \tanh \lambda_1 = \tanh(\mu_1 + \eta_{cm}) = \frac{\tanh \mu_1 + \tanh \eta_{cm}}{1 + \tanh \mu_1 \cdot \tanh \eta_{cm}}$$

$$= \frac{w_1' \cos \alpha + v_{cm}}{1 + w_1' v_{cm} \cos \alpha}$$

But, $\sin^2 \alpha = \left(\frac{v_1'}{w_1'}\right)^2 \sin^2 \theta_1$, so $(w_1')^2 \cos^2 \alpha = (w_1')^2 - (v_1')^2 \sin^2 \theta_1$.

It implies

$$v_1' \cos \theta_1 = \frac{\sqrt{(w_1')^2 - (v_1')^2 \sin^2 \theta_1} + v_{cm}}{1 + v_{cm} \sqrt{(w_1')^2 - (v_1')^2 \sin^2 \theta_1}}$$

$$v_1' \cos \theta_1 + v_{cm} v_1' \cos \theta_1 \sqrt{(w_1')^2 - (v_1')^2 \sin^2 \theta_1} = v_{cm} + \sqrt{(w_1')^2 - (v_1')^2 \sin^2 \theta_1}$$

$$v_1' \cos \theta_1 - v_{cm} = (1 - v_{cm} v_1' \cos \theta_1) \sqrt{(w_1')^2 - (v_1')^2 \sin^2 \theta_1}$$

Squaring this equation, we'll get

$$\begin{aligned} & [v_{cm}^2 \sin^2 \theta_1 \cos^2 \theta_1] (v_1')^4 - [2v_{cm} \cos \theta_1 \sin^2 \theta_1] (v_1')^3 + [1 - v_{cm}^2 (w_1')^2 \cos^2 \theta_1] (v_1')^2 \\ & + [2v_{cm} \cos \theta_1 ((w_1')^2 - 1)] v_1' + [v_{cm}^2 - (w_1')^2] = 0 \end{aligned}$$

We can find v_1' (principally) from this polynomial.

We also can claim that we can get α from equation

$$\sin \alpha = \frac{v_1'}{w_1'} \sin \theta_1$$

where v_1' and w_1' are functions of v and θ_1 .

Find θ_2 & v_2' .

$$v_2' \sin \theta_2 = \tanh \bar{\lambda}_2 = \tanh \bar{\mu}_2 = w_2' \sin \alpha$$

$$v_2' \cos \theta_2 = \tanh \lambda_2 = \tanh (-\mu_2 + \eta_{cm}) = \frac{\tanh \eta_{cm} - \tanh \mu_2}{1 - \tanh \eta_{cm} \cdot \tanh \mu_2}$$

$$= \frac{v_{cm} - w_2' \cos \alpha}{1 - v_{cm} w_2' \cos \alpha}$$

$$\text{So, } \tan \theta_2 = w_2' \sin \alpha \cdot \frac{1 - v_{cm} w_2' \cos \alpha}{v_{cm} - w_2' \cos \alpha}$$

where w_2' , α and v_{cm} are functions of v and θ_1 .

$$v_2' = w_2' \frac{\sin \alpha}{\sin \theta_2}$$

where w_2' , α and θ_2 are functions of v and θ_1 .